Zircaloy Cladding Degradation Under Repository Conditions

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Abstract

Creep, a potential degradation mechanism of Zircaloy cladding after repository disposal of spent nuclear fuel, has been investigated. The deformation and fracture map methodology has been used to predict maximum allowable initial storage temperatures to achieve a thousand year life without rupture as a function of spent-fuel history. Maximum allowable temperatures are 340°C (613 K) for typically stressed rods (70-100 MPa) and 300°C (573 K) for highly stressed rods (140-160 MPa).

Key Terms: Zircaloy, repository, creep, stress corrosion, deformation and fracture maps

Introduction

The U.S. Department of Energy's Office of Civilian Radioactive Waste Management is investigating the suitability of the tuffaceous rocks at Yucca Mountain in Nevada for the permanent disposal of high-level radioactive waste and spent fuel. Regulatory requirements specify that in general the engineered barrier system must not release more than one part in 105 per year of the inventory of each radionuclide present at 1,000 years post-closure. This release requirement holds for 10,000 years post-closure. 1,2 The Zircaloy cladding that already exists on the fuel rod may be an important barrier that contributes to the ability to meet the above release requirement. This paper presents results from an investigation conducted in support of studies designed to address the integrity of the Zircaloy cladding after disposal in a tuff repository.

Three distinct time periods can be identified during repository storage: 3

1. The high-temperature period during which the container is assumed to be unbreached, the fuel rods are surrounded by inert gas or air and no liquid water is present (0 to 90 years).

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- 2. The intermediate-temperature period when the container is assumed to be unbreached and liquid water (from breached fuel rods containing water) may be present in contact with the fuel rods (90 to 1000 years).
- 3. The lower-temperature period when the container is assumed to be breached and air, water vapor and liquid water may be in contact with the spent fuel rods (1000 to 10,000 years and beyond). Water from well J-13 is currently thought to be similar to the water that may contact the breached canister. 4 The composition of J-13 well water may be found in reference 2.

Several potential mechanisms of Zircaloy cladding degradation during dry storage have been identified such as hydride reorientation, stress corrosion cracking, creep and creep fracture. A summary discussion of potential mechanisms of degradation is given by reference 2. This paper addresses only creep/creep fracture which is considered to be the dominant degradation mechanism during time period 1.

Creep and long-term fracture damage occurs during the inert gas isolation phase of the storage (period one). Deformation and fracture map analyses have been used to predict the time-to-rupture of Zircaloy-clad-spent Light Water Reactor (LWR) fuel subjected to various storage conditions. The questions that this analysis attempts to answer are how long and under what temperature conditions the fuel can be reasonably claimed to be unfailed. In the repository, temperatures are anticipated to rise first as the repository is filled and then decrease with increasing time due to a decay in the heat generation rate of the spent fuel. In this analysis, a life fraction rule has been used to account for changing temperature and stress conditions. This life fraction rule has been combined with the results of deformation and fracture maps to predict maximum allowable temperatures to achieve a 1000 year life.

Deformation and Fracture Maps

Background

One of the probable mechanisms of failure during the inert gas isolation phase is creep rupture failure. Previous predictions of Zircaloy cladding failure have been based on direct extrapolation of existing short term data 5,6 or have assumed a single equation, like the Larson-Miller parameter 7, to describe the failure mechanisms. These methods ignore changes in the deformation and failure mechanisms as the stress and temperature conditions are altered and resulting failure times change. Such analyses are very useful for interpolation between existing data fields but often lead to erroneous results when extrapolated to regimes involving different deformation and failure phenomena.

Deformation and fracture maps enable predictions of the dominant deformation and failure mode for varyin; temperature and stress regimes, as well as varying irradiation histories. Beformation and fracture maps can be generated using the model based method in which different theoretical expressions for deformation or fracture are evaluated. The mechanism with the highest deformation rate is assumed to be the dominant mechanism. Similarly,

in the generation of the model-based fracture map, values of time-to-failure are numerically evaluated for possible fracture mechanisms, with the assumption that the mechanism with the shortest failure time controls the fracture. An empirical life fraction rule is combined with the results of the deformation and fracture maps to predict acceptable storage and disposal temperatures.

Analysis Method

The deformation and fracture map methodology has been applied in the licensing of several dry storage facilities. A detailed description of the methodology can be found in references 7,8. The present work describes modifications to dry storage deformation and fracture maps to predict cladding response for assumed repository conditions over a 1000-year period. This included the selection of a function to model the temperature decay history of the repository, refinement of the parameters that control the recovery process to better describe the region of lower repository temperatures, and incorporation of variable time steps to improve the accuracy of cumulative damage predictions at early times where the temperature is changing rapidly.

Selection of temperature decay function. The projected peak cladding temperature as a function of time in a tuff repository is shown in Figure 1^2 . This temperature history is based upon horizontal emplacement in a repository at 44 kW/acre of a canister containing 10 year old spent fuel (3.3 kW) with 12 internal fins. This temperature curve can be approximated by the equation given below, which was then incorporated into the models:

$$T(K) - T_0(K) * (time in months)^{-0.1041}$$
 (1)

Selection of melting temperature. The experimental melting temperature of beta phase Zircaloy, used in the dry storage predictions, was replaced by the theoretical melting temperature of alpha phase Zircaloy (calculated to be 1900 K). Based upon a review of the pertinent literature regarding the use of the theoretical melting temperature of alpha zirconium, it was concluded that a better description of the annealing and dislocation motion process (which occur in the alpha Zircaloy phase), could be obtained by using the theoretical alpha Zircaloy melting temperature instead of the beta Zircaloy melting temperature. All applications of the deformation and fracture map models to the repository case involve only the alpha phase. Use of the beta Zircaloy melting temperature results in a prediction that underestimates the recovery process.

Incorporation of variable time steps. Another modification involved incorporation of variable time steps to calculate the cumulative damage. Small time steps correspond to small temperature and stress changes, which improve the accuracy of cumulative damage calculations. The variable time step employed utilizes small time steps when the temperature is changing rapidly and larger steps as the rate of temperature change decreases.

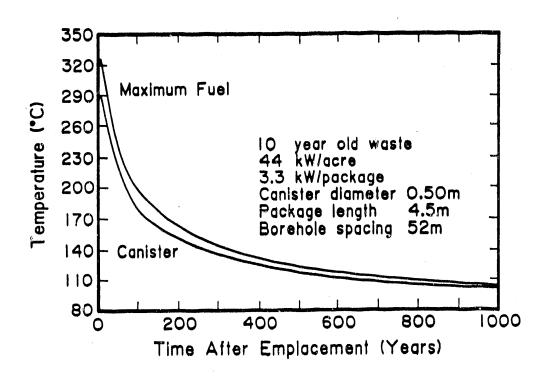


Figure 1. Peak Fuel and Canister Temperature as a Function of Time (After Rothman reference 2)

Calculations

Table 1 lists the deformation equations used in the analyses and Table 2 lists the coefficients and the symbols used in the equations. Table 3 lists the reduced deformation equations after substitution of these coefficients. Table 4 lists the fracture equations; Table 5 shows the symbols and coefficients used in these equations and Table 6 lists the reduced fracture equations.

Deformation rates were evaluated from the equations listed in Table 3. The dominant deformation mechanism is the one with the highest deformation rate. Values of time-to-failure were determined from the equations in Table 6. The mechanism with the least time to failure controls the fracture. Note that the equations in Table 6 are dependent upon the deformation rate which is calculated from the Equations of Table 3. A detailed description of the development of the equations and their use is given in references 8,10.

A life fraction rule was used to account for changing temperature and stress conditions. Temperatures are assumed to decrease with increasing time as per equation (1) after the initial warm up period. Additionally the stress in the cladding will decrease due to cladding creep and a decrease in the fission gas pressure due to decreasing temperature. Small increases due to helium gas production from actinide decay were neglected for this 1000-year analysis. This life fraction rule can be mathematically represented as follows:

$$1 = \frac{\Delta t_1}{\tau_1} + \frac{\Delta t_2}{\tau_2} + \frac{\Delta t_3}{\tau_3} + \ldots, \tag{2}$$

or

$$\sum_{i=1}^{N} \frac{\Delta t_i}{r_i} - 1 \tag{3}$$

where Δt_i is the time spent at the ith temperature, and τ_i is the time required to fracture a specimen under isothermal, isostress testing conditions. When the cumulative damage fraction $(\Delta t_i/\tau_i)$ reaches one, the fuel rod is assumed to have failed.

Results

It is assumed that the fuel rods will have a temperature decay history as described by equation 1. To determine the maximum allowable temperatures, a computer program was developed. This program uses the deformation and fracture map methodology in conjunction with the life fraction rule to predict behavior in the repository. The results are depicted in Figure 2, which is a plot of maximum allowable peak temperature as a function of clad hoop stress for 10 year old spent fuel (10-years in wet storage) and an initial clad stress of 70 MPa. This maximum allowable temperature is the maximum peak temperature (for a given initial clad stress) following the temperature decay rate shown in Figure 1 that yields a cumulative damage fraction of less than one after an elapsed time of 1000 years.

Table 1Deformation Equations

High Temperature Climb	$\dot{\epsilon}_{HT} = A_1 \cdot D_{o1} \cdot \exp(-Q_1/RT) \cdot \frac{Eb}{KT} \cdot (\sigma/E)^n$
Low Temperature Climb	$\dot{\epsilon}_{LT} = 50 \cdot A_1 \cdot D_{oc} \cdot \exp(-Qc/RT) \cdot \frac{Eb}{KT} \cdot (\sigma/E)^{n+2}$
Grain Boundary Sliding	$\dot{\epsilon}_{GBS} = A_{GBS} \cdot D_{ogb} \cdot exp(-Qgb/RT) \cdot \frac{Eb}{KT} \cdot (b/d)^3 \cdot (\sigma/E)^2$
Nabarro-Herring	$\dot{\epsilon}_{\text{NH}} = A_{\text{NH}} \cdot D_{\text{ol}} \cdot \exp(-Q_1/RT) \cdot \frac{Eb}{KT} \cdot (b/d)^2 \cdot (\sigma/E)$
Coble	$\dot{\epsilon}_{\text{CO}} = A_{\text{CO}} \cdot D_{\text{Ogb}} \cdot \exp(-Qgb/RT) \cdot \frac{Eb}{vm} \cdot (b/d)^3 \cdot (\sigma/E)$

Table 2

Symbols and Coefficient Values for Deformation Equations

```
- strain rate
σ
        = stress
Т
       - temperature
       - gas constant - 8.3144 J/mole-K
R
k
       = Boltzmann's constant = 1.38 \times 10^{-29} \text{ MJ/K}
       - Burger's vector - 3.23 \times 10^{-10} \text{ m}
Ъ
       = grain size = 5.0 \times 10^{-6} m
       - melting temperature of Zircaloy - 2125K
       - Young's Modulus (11.81 - 14.59 T/T_m) x 10<sup>4</sup> MPa, T_m/T > 4.06 for T < 865°C (11.09 - 11.61 T/T_m) x 10<sup>4</sup> MPa, T_m/T < 4.06
       = stress exponent = 5
n
Q
       - activation energy
       = 250 \text{ kJ/mol}
Q_{gb} = 175 \text{ kJ/mol}
       - 180 kJ/mol
Q_c
       - diffusivity coefficient
D_{o}
D_{01} = 2.00 \times 10^{-4} \text{ m}^2/\text{s}
D_{ogb} = 3.89 \times 10^{-6} \text{ m}^2/\text{s}
D_{\rm oc} = 2.26 \times 10^{-6} \, \text{m}^2/\text{s}
       - strain rate coefficient
       -7.4 \times 10^{-7}
Αı
A_{CO} = 5.362 \times 10^2
A_{GBS} = 8.85 \times 10^6
A_{NH} = 0.92
A_{GBL} = 5 \times 10^2
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Table 3

Reduced Deformation Equations

High Temperature Climb $\ln \epsilon_{\rm HT} = 5 \cdot \ln(\sigma/E) + 55.75 \cdot 14.15 \left(T_{\rm m}/T\right) + \ln(T_{\rm m}/T) + \ln(E/10^4)$

Low Temperature Climb $\ln \epsilon_{LT} = 7 \cdot \ln(\sigma/E) + 55.18 \cdot 10.19 (T_m/T) + \ln(T_m/T) + \ln(E/10^4)$

Grain Boundary Sliding $\ln \epsilon_{GBS} = 2.\ln(\sigma/E) + 20.74 - 9.9036(T_m/T) + \ln(T_m/T) + \ln(E/10^4)$

Nabarro-Herring $\ln i_{NH} = \ln(\sigma/E) + 18.25 - 14.15(T_m/T) + \ln(T_m/T) + \ln(E/10^4)$

Coble $\ln \dot{\epsilon}_{CO} = \ln(\sigma/E) + 11.03 - 9.9036(T_m/T) + \ln(T_m/T) + \ln(E/10^4)$

Table 4

Fracture Equations

Theoretical Shear Strength
$$\sigma_{th} = \frac{E \gamma_f}{a_o} \approx \frac{E}{10}$$

Transgranular Fracture
$$t_{f}^{TG} = \epsilon_{n} + \left(\frac{1}{4.937}\right) \left(\frac{n}{n-1}\right) \ln \frac{0.38}{f_{v}^{1/2}} - 1 \epsilon^{-1}$$

Triple-Point Cracking
$$t_f^{TP} = \frac{f}{Ed} (\frac{\sigma}{E})^{-1} \dot{\epsilon}^{-1}$$

Cavitation-Diffusional Growth
$$t_f^{CD} = \frac{2.525 \times 10^{-3} \ell^3}{\delta D_{ogb} b^2 \exp \frac{-Q_{gb}}{RT}} \left(\frac{k_T}{Eb}\right) \left(\frac{\sigma}{E}\right)^{-1}$$

Cavitation--Power Law Growth
$$t_{f}^{CP} = \frac{(1 - 0.78 P_{o} l_{o})}{4.87} \dot{\epsilon}^{-1}$$

Table 5

Symbols and Coefficient Values for Fracture Equations

- tf time to fracture
- ϵ_n hole nucleation strain 0.08
- f_v volume fraction of intragranular inclusions 0.025
- $P_{\rm o}$ average particle diameter 100 Å
- $\ell_{\rm o}$ particle spacing along boundary 2.0 x 10⁻⁶ m
- $\zeta = \epsilon_{\rm GBS}/\epsilon$ contribution of grain boundary strain rate to total strain rate = 0.2
- ℓ = average cavity spacing = 2.6 x 10^{-6} m (6 per grain segment)
- δ = width of grain boundary = 1.6 x 10⁻⁸ m (50 Burger's vectors)
- $_{\rm f}$ free surface energy created by fracture 35 J/m²
- a₀ lattice spacing

Table 6

Reduced Fracture Equations

Transgranular Fracture

$$\ln t_f^{TG} = 1.797 - \ln \epsilon$$

Triple-Point Cracking

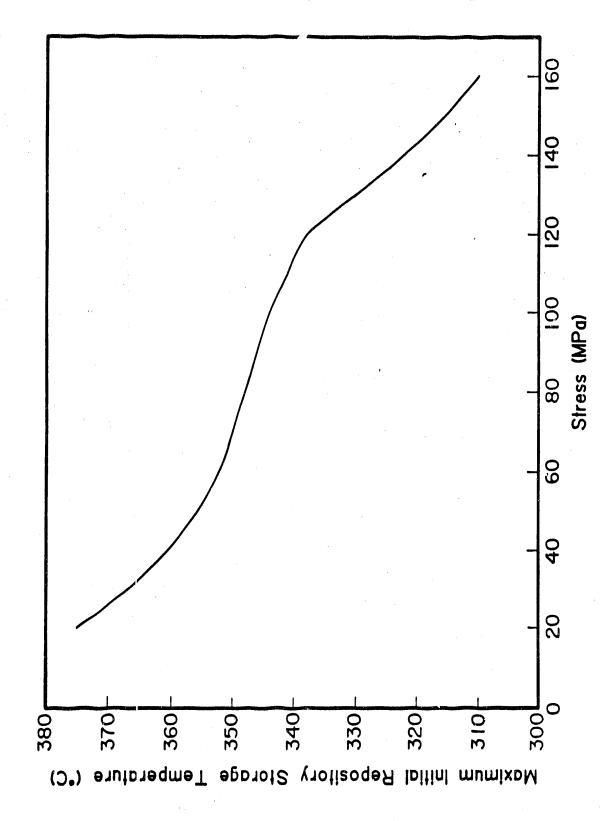
$$\ln t_f^{TP} = -5.655 - \ln \epsilon - \ln \frac{\sigma}{E} - \ln \frac{E}{10^4}$$

Cavitation -- Diffusional Growth

$$\ln t_f^{CD} = + 4.15 - \ln \epsilon_{gbs} + \ln \frac{\sigma}{E}$$

Cavitation -- Power Law Growth

$$\ln t_f^{CP} = -1.587 - \ln \epsilon$$



Predicted Peak Repository Temperature as a Function of Initial Fuel Rod Stress Figure 2.

For times of 1000 years and beyond the damage accumulation rates are so low (because of the low repository temperatures) that calculations beyond 1000 years are not necessary.

For typically stressed rods (70 -100 MPa) with low fission gas release, initial temperatures of 340°C would be acceptable. However for high burnup rods with large fission gas release that results in highly stressed rods (140-160 MPa), initial temperatures should not exceed 300°C. It should be noted that the predicted maximum allowable temperature is a function of initial clad stress (fuel rod history), age of spent fuel (cumulative time spent in wet storage) and anticipated temperature history of the repository. The calculations are conservative for spent fuel that is younger than 10 years post discharge, but nonconservative for older fuel rods. This dependence upon time since discharge results from the higher rates of temperature decay in younger spent fuel versus older spent fuel. Additionally, the temperature history curve shown in Figure 1 does not include the initial region of temperature increase that occurs during filling of the repository.

Conclusions

Zircaloy cladding is potentially an effective barrier to the release of radionuclides from spent fuel in a repository. After making some assumptions about Zircaloy material behavior, preliminary models have been developed to address maximum storage temperature limits. The following conclusions resulted from this preliminary study:

Based on a deformation and fracture map analysis, predictions of maximum allowable temperatures for cladding stresses from 20 MPa to 160 MPa have been made for times upto 1000 years post-emplacement. Maximum allowable temperatures are 340°C (613 K) for typically stressed rods (70 -100 MPa) and 300°C (573 K) for highly stressed rods (140 -160 MPa).

Sponsorship

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Appendix A

This report does not use any information from the Reference Information Base nor contain any candidate information for the Reference Information Base or the Site and Engineering Properties Data Base (SEPDB).

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