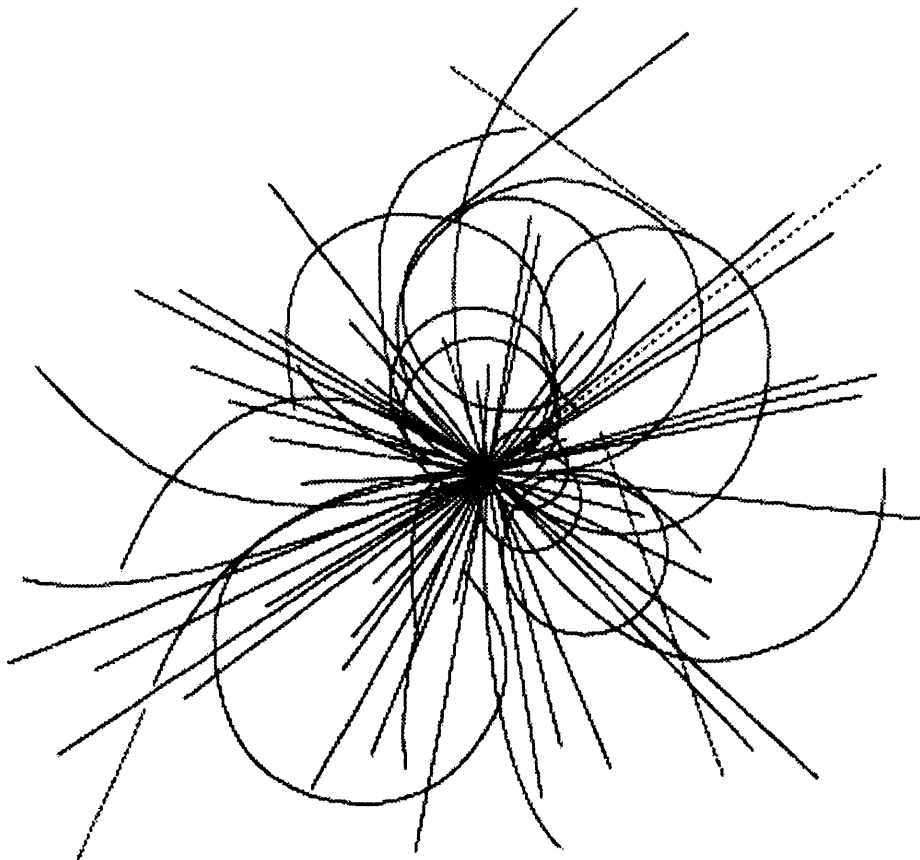


SSCL-Preprint-446
May 1993
Distribution Category: 400

Q. Shu
K. Yu
W. Clay
J. Maddocks
G. Morales
J. Zbasnik

Thermal Model and Associated Novel Approach for Synchrotron Radiation Liner with End Cooling



RECEIVED
JUL 19 1993
OSTI

Superconducting Super Collider
Laboratory

Disclaimer Notice

This report was prepared as an account of work sponsored by an agency of the United States Government. Neither the United States Government or any agency thereof, nor any of their employees, makes any warranty, express or implied, or assumes any legal liability or responsibility for the accuracy, completeness, or usefulness of any information, apparatus, product, or process disclosed, or represents that its use would not infringe privately owned rights. Reference herein to any specific commercial product, process, or service by trade name, trademark, manufacturer, or otherwise, does not necessarily constitute or imply its endorsement, recommendation, or favoring by the United States Government or any agency thereof. The views and opinions of authors expressed herein do not necessarily state or reflect those of the United States Government or any agency thereof.

Superconducting Super Collider Laboratory is an equal opportunity employer.

**Thermal Model and Associated Novel Approach
for Synchrotron Radiation Liner with End Cooling***

Q. Shu, K. Yu, W. Clay, J. Maddocks, G. Morales, and J. Zbasnik

Superconducting Super Collider Laboratory[†]
2550 Beckleymeade Ave.
Dallas, TX 75237

May 1993

*Presented at the 1993 IEEE Particle Accelerator Conference on May 17-20, Washington, D.C.

[†]Operated by the Universities Research Association, Inc., for the U.S. Department of Energy under Contract No. DE-AC35-89ER40486.

MASTER

Thermal Model and Associated Novel Approach for Synchrotron Radiation Liner with End Cooling

Quan-Sheng Shu, Kun Yu, Wayne Clay
Jim Maddocks, Gilberto Morales, and Jon Zbasnik
Superconducting Super Collider Laboratory*
2550 Beckleymeade Ave., Dallas, TX 75237 USA

Abstract

An end-conductive cooling approach has been developed to reduce the radial space budget of a synchrotron radiation liner to permit the maximum possible liner tube inner diameter (ID). A thermal model has also been developed to analyze the thermal performance of such liners. This approach is found to be acceptable for a liner in a 5-m-long quadrupole magnet and 3-m-long spool piece, but not for a longer 15-m dipole. The heat transfer and temperature distribution were calculated respectively along the axis of two different liner models: 20 K and 80 K liner with different thicknesses (0.5 – 2 mm) of liner tubes and different emissivities (0.05 – 0.3) of liner surface for a variety of magnets. The thermal model is also applied to the case of an 80 K liner connected directly to a 4 K beam position monitor (BPM). In order to utilize the end cooling, a good thermal joint and a compact heat exchanger are designed.

I. INTRODUCTION

A uniform and maximum possible liner inner diameter (ID) is needed due to: (1) particle beam commissioning, (2) particle beam dynamic stability, and (3) safety margin of impedance. However, the maximum liner ID is constrained by: (1) the available magnet beam tube inner diameter (ID), and (2) the minimum liner radial space. Using regular cooling, the minimum liner radial space is 6 mm, and using end-conducting cooling, the radial space needs to be 3.5 mm. The 80 K synchrotron radiation liner prototype was designed to be tested at the Superconducting Super Collider Laboratory (SSCL) Accelerator System String Test (ASST) facility. In the case of the 80 K ASST liner, the 25.3-mm design was chosen for the maximum liner ID.

Since the magnet quench-induced Lorentz pressure on a CQM liner is much smaller than that on a CDM liner, the pure copper tube was chosen for the CQM liner material. The RRR and thickness of the copper tube must be of sufficient value due to both requirements: (1) resistance wall: conductivity \times Thickness $> 2 \times 10 \Omega^{-1}$, and (2) conducting heat transfer requirement. However, the RRR and thickness shall not be too large in order to reduce the Lorentz pressure. This paper will focus on a thermal model used to predict the thermal performance of an end-cooling liner for different cases.

II. THERMAL MODEL FOR END-CONDUCTIVE COOLING

An end-conductive cooling approach for the Spool Piece and CQM is shown in Figure 1. The 80 K GHe flows through a compact heat exchanger located at each end of the liner tube outside of the CQM cold mass. The rest of the liner tube is refrigerated by thermal conduction. A compact heat exchanger and a good thermal conducting joint is designed to utilize the end-cooling approach and to assure an easy assembly.

A thermal model to analysis the end conductive cooling was developed by Q. S. Shu and K. Yu [1] [2], assuming: Q_s synchrotron radiation, 0.14 W/m; Q_L (heat leak through support)/2L; Q_r (heat leak by radiation)/2L; L, half length of the CQM or Spool Piece; A, the cross section area of the liner tuber; $\lambda(T)$, the heat conductivity; $\lambda(80)_{Cu} \approx 5.50$ W/(cm.K); $\lambda(80)_{snl\ stl} \approx 0.045$ W/(cm.K); ϵ_1, ϵ_2 the emissivity.

$$Q_T = \sigma A (T_4^4 - T_2^4) \epsilon_1 \epsilon_2 / (\epsilon_1 + \epsilon_2 - \epsilon_1 \epsilon_2)$$

$$dQ_c = (Q_s - Q_L - Q_r) dX$$

$$\text{if } q = (Q_s - Q_L - Q_r)$$

$$Q_x = -\lambda(T) A dT / dX$$

$$\text{and } Q_{x+dx} = -\lambda(T) A d [T + (dT / dX) dX] / dX$$

$$= -\lambda(T) A [dT / dX + (d^2T / dX^2) dX]$$

$$\text{we know } Q_{x+dx} = dQ_c + Q_x$$

$$d^2T / dX^2 = -q / [\lambda(T) A]$$

$$T(X) = -\{q / [2\lambda(T) A]\} X^2 + C_1 X + C_2$$

boundary conditions:

$$T(X) /_{x=L} = 80 \text{ K}; \quad T(X) /_{x=-L} = 80 \text{ K}$$

we have,

$$T(X) = -\{q / [2\lambda(T) A]\} X^2 + \{q / [2\lambda(T) A]\} L^2 + 80 \quad (1)$$

Using the model: 1) The temperature distribution along the liners as functions both of the emissivities and of the tube thicknesses were calculated. 2) The maximum ΔT could be less than 5 K for Spool Piece liner, and 10 K for CQM. 3) A temperature difference between the Spool Piece pipe ends and the middle of the liner is 2 K when copper layer of 2 mm is used and 10 K when copper layer of 0.5 mm was used. 4) For CQM, ΔT of 6 K is obtained with copper layer of 2 mm, and

*Operated by Universities Research Association, Inc., for the U.S. Department of Energy under Contract No. DE-AC35-89ER40486.

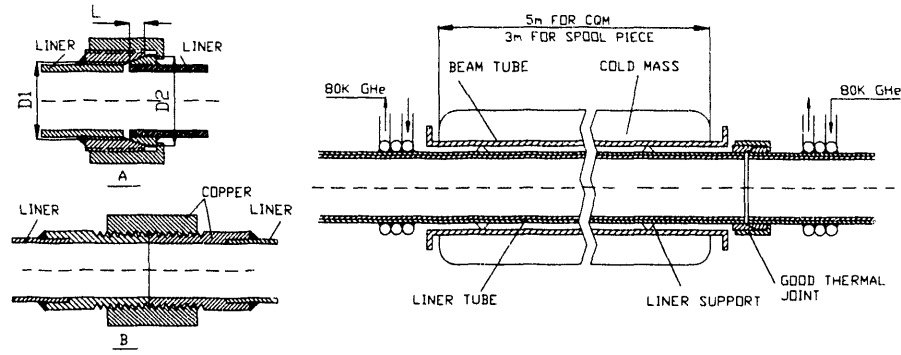


Figure 1. A schematic drawing of a liner with the end conducting cooling.

26 K with copper layer of 0.5 mm. 5) The correction of the effect of the magnetic field on copper thermal conductivity is considered. Figures 2, 3, and 4 show some of the calculated results.

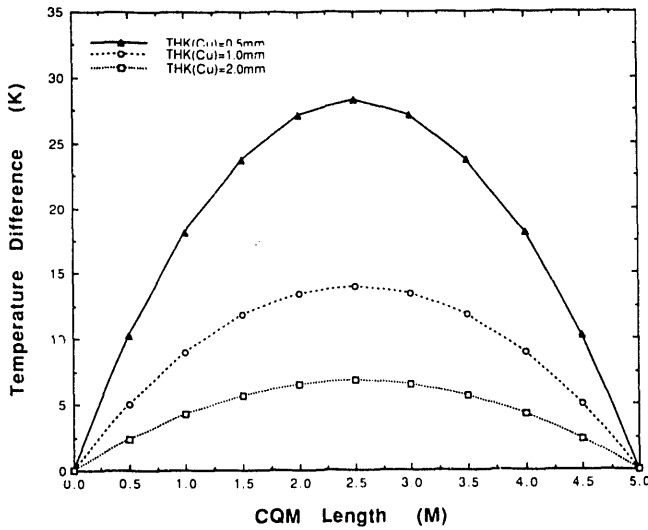


Figure 2. Temperature distribution of 80 K liners as function of liner tube thicknesses.

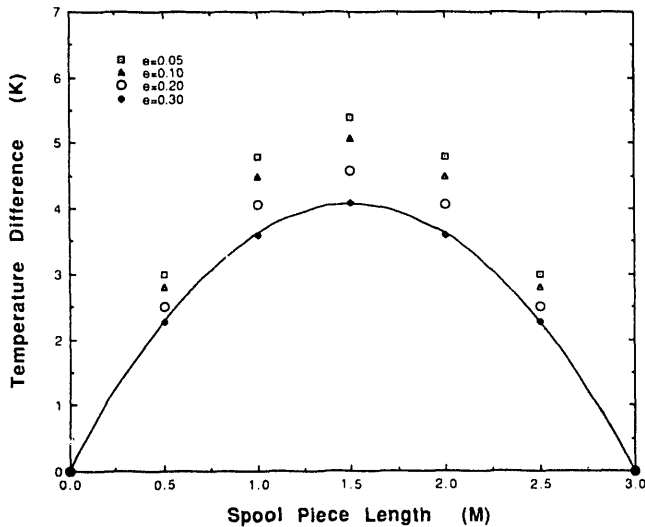


Figure 3. Temperature distribution of 80 K liners as function of liner surface emissivities.

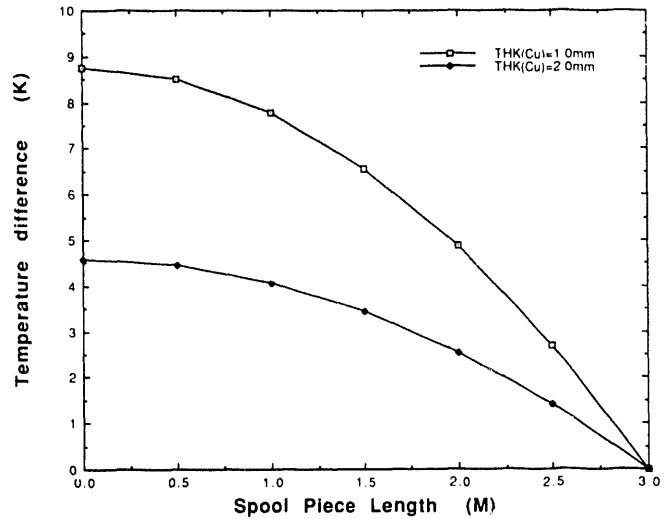


Figure 4. Temperature distribution of 20 K liners with only one end cooling.

III. THERMAL MODEL FOR 80 K LINER WITH A 4 K BPM

If a 4 K BPM is used, the 80 K liner end-conductive cooling becomes more complicated, as shown in Figure 5. To reduce the heat leak through the copper liner tube from 80 K to 4 K BPM, a 10–20-cm-long piece of stainless steel tube is insert between the BPM and copper liner tube. The synchrotron radiation, the heat leak from liner to BPM, the heat leak from liner to 4 K beam tube, and the heat exchange between the liner and the 80 K GHe must meet the law of conservation of energy.

First, we calculate a temperature distribution along the stainless steel Liner tube; when $0 \leq X \leq L$, we have,

$$T_s(X) = -qX^2 / (2\lambda_s A_s) + S_1 X + S_2 \quad (2)$$

The temperature distribution along the copper tube shall satisfy Eq.(2) if $L \leq X \leq L_0$.

$$T_c(X) = -qX^2 / (2\lambda_c A_c) + C_1 X + C_2 \quad (3)$$

Eqs. (2) and (3) must meet the following boundary conditions:

$$T_s(0) = 4$$

$$T_c(L_o) = 80$$

$$\lambda_c A_c \left\{ \frac{\partial T_c(X)}{\partial X} \right\} \Big|_{x=L} = \lambda_s A_s \left\{ \frac{\partial T_s(X)}{\partial X} \right\} \Big|_{x=L}$$

$$T_s(L) = T_c(L)$$

The C_1 , C_2 , S_1 , and S_2 can be determined:

$$S_2 = 4$$

$$S_1 = \left[\lambda_c A_c / (\lambda_s A_s) \right] \left[80 + qL_o^2 / (2\lambda_c A_c) - \right.$$

$$\left. qL^2 / (2\lambda_c A_c) + qL^2 / (2\lambda_s A_s) - 4 \right]$$

$$/ \left[\lambda_c A_c L / (\lambda_s A_s) - L + L_o \right]$$

$$C_1 = \left[80 + qL_o^2 / (2\lambda_c A_c) - \right.$$

$$\left. qL^2 / (2\lambda_c A_c) + qL^2 / (2\lambda_s A_s) - 4 \right]$$

$$/ \left[\lambda_c A_c L / (\lambda_s A_s) - L + L_o \right]$$

$$C_2 = 80 + qL_o^2 / (2\lambda_c A_c) - L_o \left\{ \left[80 + qL_o^2 \right. \right.$$

$$\left. / (2\lambda_c A_c) - qL^2 / (2\lambda_c A_c) \right.$$

$$\left. + qL^2 / (2\lambda_s A_s) - 4 \right] / \left[\lambda_c A_c L / (\lambda_s A_s) \right.$$

$$\left. - L + L_o \right\}.$$

Using Eqs. (2) and (3) the temperature distribution can be calculated and is shown in Figure 6.

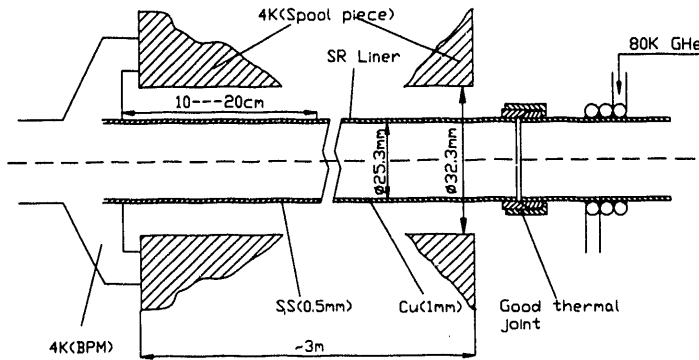


Figure 5. A schematic drawing of an 80 K liner with a 4 K BPM.

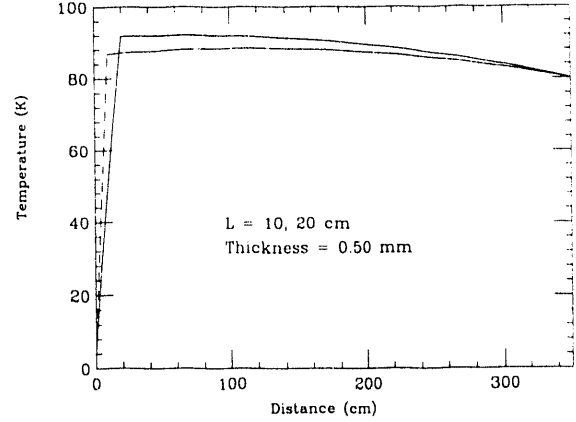


Figure 6. Temperature distribution along an 80 K liner with a 4 K BPM.

IV. GOOD THERMAL CONDUCTING JOINT

A good thermal conducting joint concept, as shown in Figure 1, was proposed by Q. S. Shu and K. Yu. The joint makes liner assembly easier. Assume synchrotron radiation of a quadrupole to be 1 W. The heat transferred at each liner end is 0.5 W. If the pressure on the Cu-Cu machined contact is 7 MPa, thermal conductance of the contact (at temperature range 5–25K) is $h(T) = 0.13 T^{-1}$ (W / cm² K). The temperature across the joint ΔT (at 80 K) ≤ 1 K.

V. COMPACT HEAT EXCHANGER

To make end-cooling work, a compact heat exchanger with a length of less than 5 cm was developed. The total heat to be transferred by the heat exchanger is $Q=2$ W. Design parameters used were: Copper cooling tube ID=0.25 cm, mass flow rate of the 80 K GHe = $dM/dt = 0.25$ g/s, temperature increase of GHe is ΔT , $Re = GD/\eta = 31812$, $Pr = \eta C_p/\lambda = 0.357$, $h = 0.023 C_p G^{0.8} \eta^{0.2} / (Pr^{0.6} De^{0.2}) = 0.0345$. If three turns are used, $L = 28.75$ cm, $\Delta T < 1$ K.

VI. ACKNOWLEDGMENTS

The authors sincerely thank A. Yucel, D. Clark, D. Martin, and W. Turner of the SSCL for their contribution to the work.

VII. REFERENCES

- [1] Q. S. Shu, Status Report on the ASST Liner System Design, SSCL-N-805, November, 1993.
- [2] 80 K ASST liner Design Report, prepared by Q. S. Shu, (in preparation).

END

**DATE
FILMED**

9 / 15 / 193

